

UTSR Fellowship - 2012



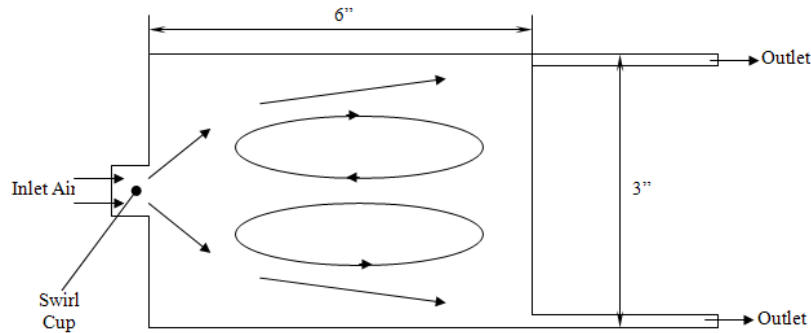
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Outline

- Problem description and Objective
- Operating condition
- CFD model
- Spray model
- Results
- summary

Problem Description and objective



Problem description

- Spray cup with radial-inflow air swirler
- A pressure swirl atomizer injected fuel at the bottom of the spray cup exposing the spray directly to the incoming swirling air
- The air and fuel from the swirl cup discharged into a 10.0" inch long 3.00" by 3.00" square tube.

Objective

Investigate different approaches for predicting spray flame location and NO_x emissions.

Operating condition

Fuel: $C_{12}H_{23}$ as a surrogate for jet-A fuel

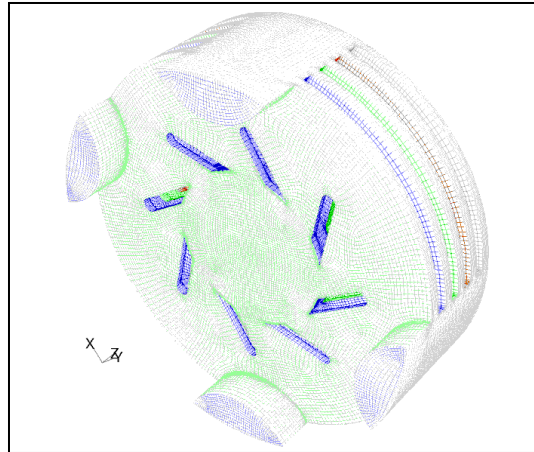
Non reacting case : Fixed pressure drop (4%)

Reacting case: Fixed mass flow rate

Type of gas : Incompressible ideal gas

CFD model

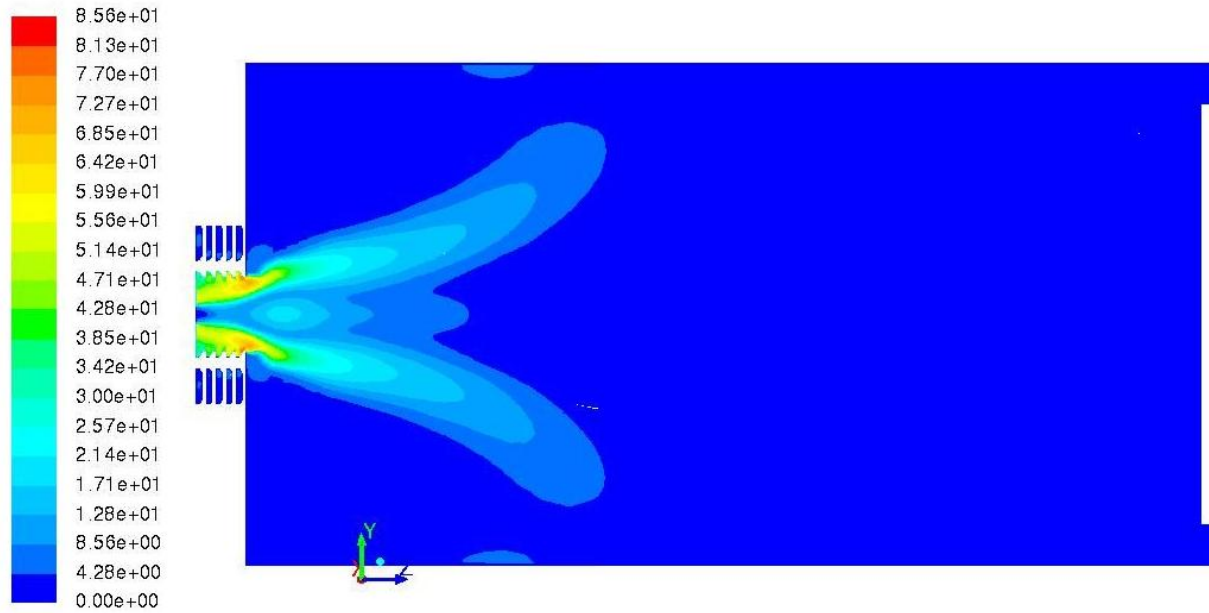
- Structured multi-block grid system that was generated using the GridPro
- The grid system contained a little over $2.0 \cdot 10^6$ hexahedral cells



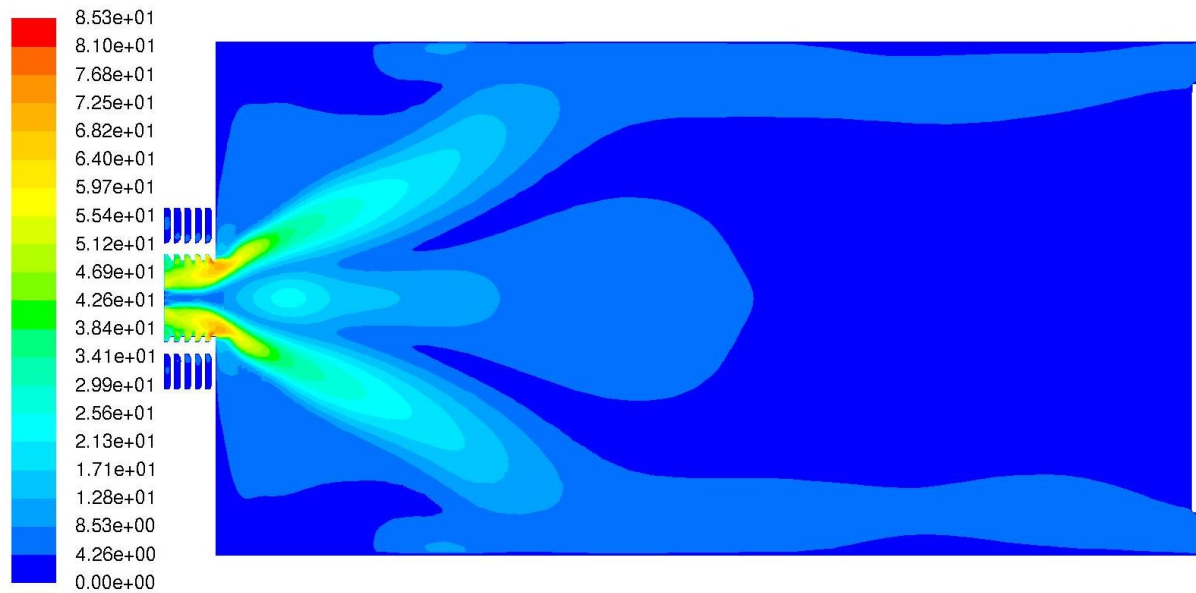
- Turbulence model : realizable k-epsilon model
- Combustion model: Flamelet model
- Reaction mechanism: 31 step reaction with 21 species

Spray Model

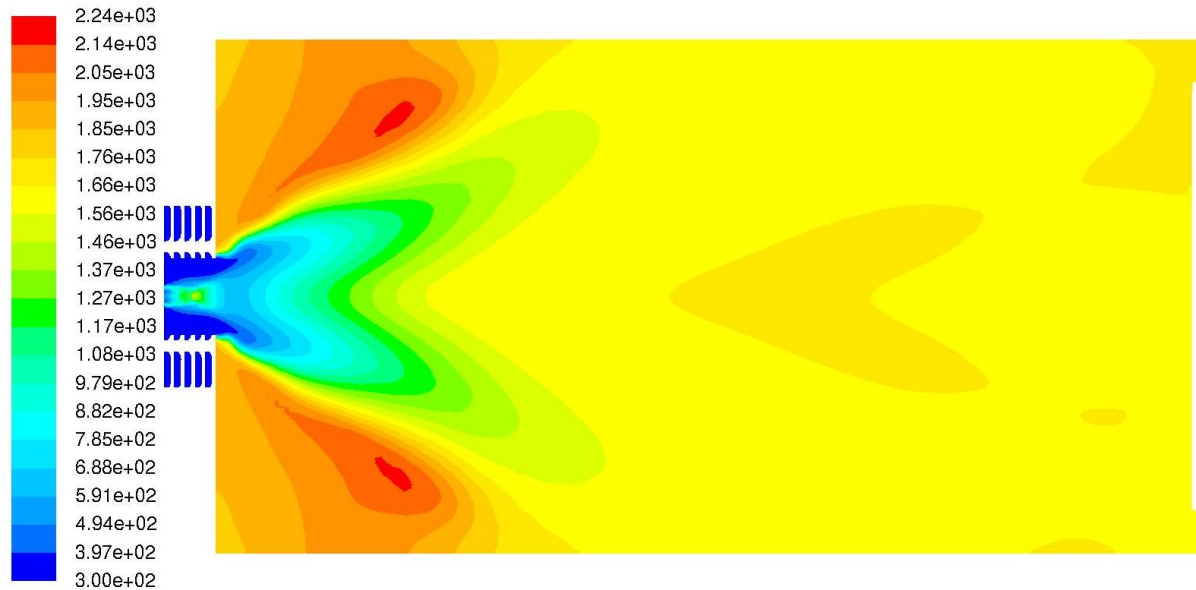
- Model**: Discrete phase model
- Particle Type** : Droplet
- Drople Size Distribution**: Rosin-Ramler distribution
- Injector**: Hollow conical spray
- Spray angle**: 80



Reacting velocity field

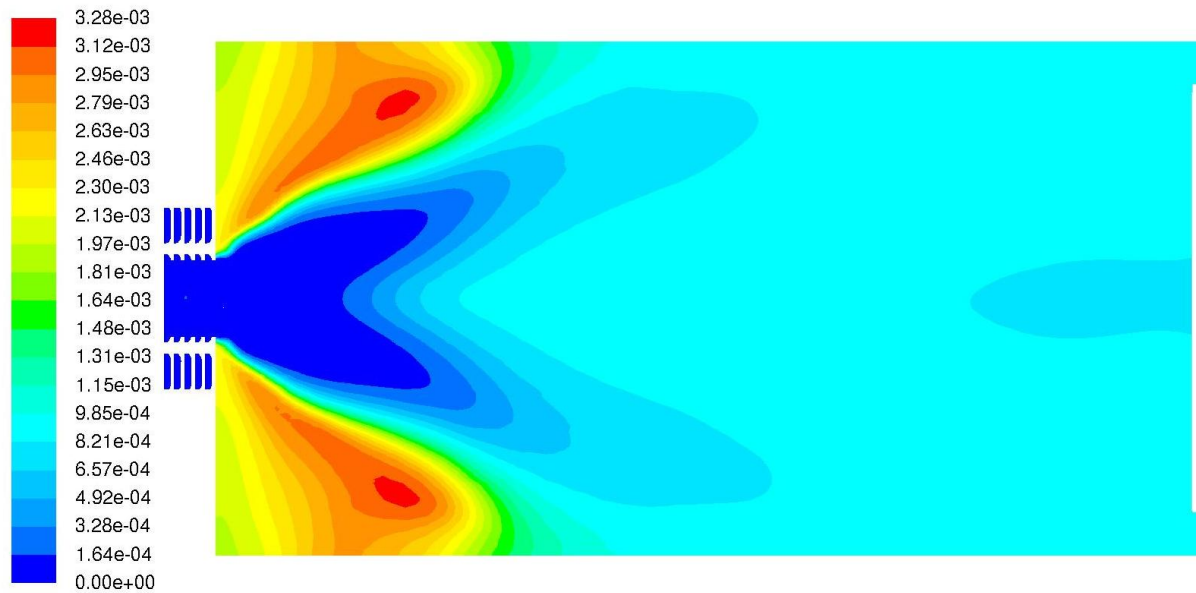


Temperature

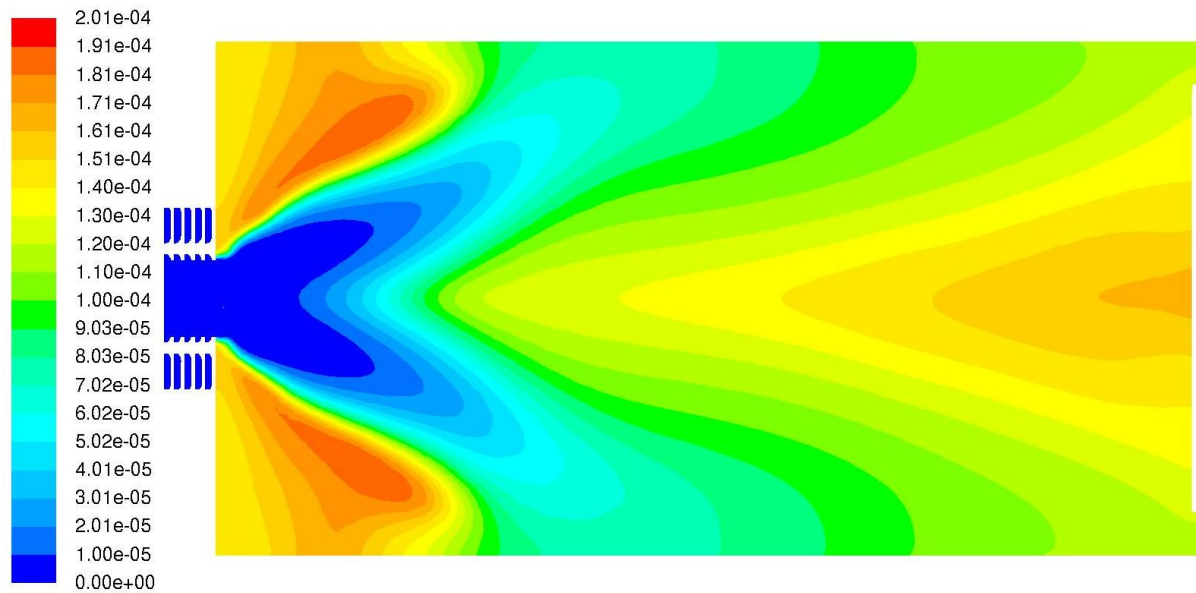


Maximum temperature: 2240 K
Exit temperature: 1638 K

Mass fraction of OH – Unsteady Flamelet Results



Mass fraction of NO – Unsteady Flamelet Results



Summary

- Strongly swirling flow downstream of the cup and swirl induced vortex breakdown that stabilizes the flame.
- Temperature profile for the reacting flow shows the maximum temperature is 2240 K and exit temperature is 1638 K with the burning zone positioned downstream of the cup
- An unsteady flamelet model was used as a post processing step to estimate NO emissions

References

1. Cai, J., Jeng, S.-M., Steinthorsson, E., “Experimental and Numerical Investigation of a Macro-Laminated Radial Swirler”, AIAA 2003-0826.
2. Tacina, R., Wey, C., Laing, P., Mansour, A., “Sector test of a low NO_x, lean-direct-injection, multipoint integrated module combustor concept,” ASME Turbo Expo 2002.