

# Development of a Brayton Bottoming Cycle using Supercritical Carbon Dioxide as the Working Fluid

by

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# Introduction

Inherent in any power cycle is the ultimate rejection of a significant amount of thermal energy, which exits the system either by heat transfer in a closed cycle, or via thermal transport by exhaust gases in an open cycle. All heat that is rejected to the environment is heat that is not being used to generate power, so needless to say, a significant amount of effort is made to make as much use of this waste heat as possible. One method of recovering this waste heat is to integrate a secondary, or bottoming cycle into the system that essentially uses the waste heat of the primary, or topping cycle, as its heat source. The use of a bottoming cycle boosts the overall plant efficiency, meaning that more usable power is generated using the same amount of fuel.

By far the most commonly encountered bottoming cycle today is the Rankine cycle, using steam as the working fluid. The Rankine cycle is very well researched, and is used worldwide, both in topping and bottoming cycle applications. Recently, however, a number of research teams have begun to look for an alternative to the classic steam cycle, with a growing focus on Brayton cycles using supercritical carbon dioxide (sCO<sub>2</sub>) as the working fluid.

Researchers claim that an sCO<sub>2</sub> power cycle could potentially exhibit a higher thermal efficiency than steam cycles when operating between the same maximum and minimum cycle temperatures. In addition, the high energy density of sCO<sub>2</sub> suggests that the size requirements for the turbomachinery used in an sCO<sub>2</sub> power cycle could potentially be much lower than those used in steam cycle generation, which may result in lower capital costs.

As a result, a Brayton power cycle using sCO<sub>2</sub> as the working fluid has the potential to replace the Rankine cycle in both topping cycle and bottoming cycle applications. To date, most research in the field has been dedicated to the use of sCO<sub>2</sub> as the primary power cycle in nuclear applications. Additional research is being conducted regarding the use of an sCO<sub>2</sub> cycle in other

primary cycle applications, but relatively little research has been aimed toward developing an sCO<sub>2</sub> cycle that is well-suited to bottoming cycle applications.

## Project Scope

The primary focus of this project at the start was to verify claims that an sCO<sub>2</sub> Brayton cycle would yield a marginally higher thermal efficiency than a Rankine steam cycle while operating between the same minimum and maximum temperatures. Three commonly proposed Brayton cycles were to be modeled and optimized, and their thermal efficiencies were to then be compared to a Rankine steam cycle operating under the same conditions. As will be discussed, the three proposed cycles turned out to be unsuitable for a bottoming cycle application, and were scrapped. At this point, the project scope evolved to include designing and modeling new potential cycle configurations, with the goal of creating a new cycle configuration that maximizes both efficiency and waste heat recovery. In all, nearly thirty different cycle configurations were conceived of, modeled, and evaluated. While most of these cycles turned out to be unsuitable for further development, three of the modeled cycles showed enough potential to be included in this report.

## Benchmarking

In order to gauge the performance of each cycle configuration, two modern combined cycles were selected as benchmarks. The two cycles chosen represent two opposite ends of a broad spectrum of combined cycles, ranging from low power, low temperature applications to high power, high temperature applications.

The first benchmark cycle is the Siemens H Class gas turbine. This cycle employs an internal combustion turbine fueled by natural gas as the topping cycle, with a 3 pressure level,

reheat steam cycle as its bottoming cycle. This combined cycle system was chosen for its record-breaking efficiency and power output, and it represents the high power, high temperature side of the spectrum. The other benchmark cycle is the GE LM6000, which also employs a natural gas topping cycle paired with a 2 pressure, non-reheat steam bottoming cycle, which represents the low temperature, low power side of the spectrum.

The modeling of each sCO<sub>2</sub> cycle used the topping cycle exhaust properties to predict the power output of the sCO<sub>2</sub> cycle if it were to be implemented as a bottoming cycle on the benchmark topping cycle. The power output of the sCO<sub>2</sub> bottoming cycle can then be compared to the power output of the benchmark bottoming cycle in order to gauge its relative performance. Shown below are the most important properties of each benchmarked combined cycle.

**Table 1. Key properties of benchmark combined cycles**

<b>Benchmark Combined Cycle</b>	<b>Siemens H class</b>	<b>GE LM6000-PH</b>
Topping Cycle Exhaust Flow Rate	820 kg/s	138.8 kg/s
Topping Cycle Exhaust Temperature	625 °C	471 °C
Bottoming Cycle Shaft Power	195 MW	14 MW

Source: Manufacturer's Data <sup>[1][2]</sup>

## Design Parameters, Constraints, and Assumptions

For the purposes of this project, some important assumptions were made in order to successfully model multiple cycles and compare their performance. Unless otherwise noted, the following assumptions were made for each case.

## Thermodynamic Fluid Properties

There are assumptions made in “textbook” calculations which cannot be justified while analyzing sCO<sub>2</sub> Brayton Cycles.

### 1. Ideal gas behavior

A real gas may be approximated as an ideal gas only at relatively low pressures and high temperatures. Generally, it is an accurate assumption if

$$P_r = \frac{P}{P_{cr}} \ll 1 \text{ or } T_r = \frac{T}{T_{cr}} > 2$$

Where  $P_{cr}$  is the pressure at the critical point and  $T_{cr}$  is the temperature at the critical point

The cycles discussed in this report operate within the range of

$$36.9^\circ\text{C} (310 \text{ K}) < T < 604.9^\circ\text{C} (878 \text{ K})$$

$$5000 \text{ kPa} < P < 27600 \text{ kPa}$$

The critical point of CO<sub>2</sub> is

$T_{cr}$	304.19 K
$P_{cr}$	7380 kPa

So the range of relative temperature and pressure is

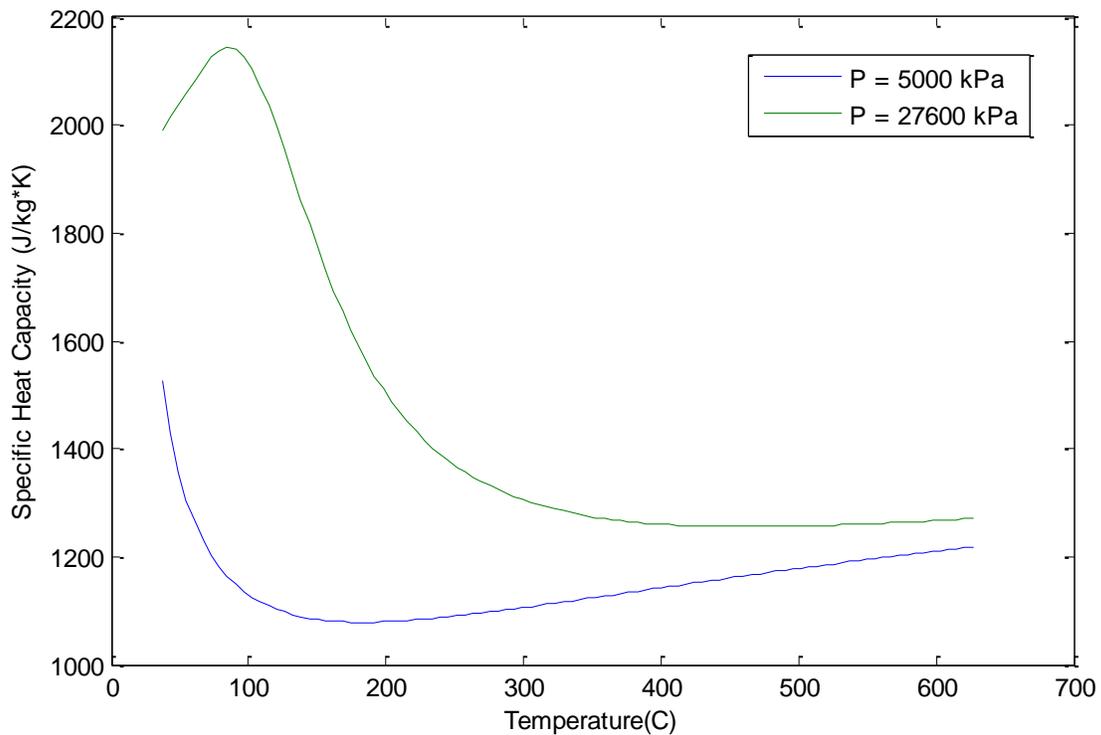
$$1.02 < T_r < 2.89$$

$$0.678 < P_r < 3.74$$

which does not justify the use of the ideal gas assumption.

### 2. Constant specific heat capacity

Specific heat capacity is a very important property of gas, particularly when analyzing heat transfer and power cycles. A common assumption is that the specific heat of a gas remains constant through small changes in pressure and temperature. This assumption is not valid for a supercritical fluid, which is shown graphically in Figure 1.



**Figure 1. Specific heat capacity of sCO<sub>2</sub> vs Temperature, calculated using NIST REFPROP thermodynamic properties database**

The heat capacity varies substantially with temperature and with pressure, meaning the constant specific heat capacity assumption is invalid. Instead, fluid properties such as temperature and enthalpy are calculated directly using the NIST Reference Fluid Thermodynamic and Transport Properties Database (REFPROP) version 9.0.

### **Operating Constraints**

For the purposes of this project, some operating conditions were imposed in order to keep costs low and to create a realistic model. The maximum operating pressure is limited to 276 bar

(4000 PSI) in order to avoid using excessively thick piping and flanges. The maximum operating temperature is limited by the exhaust temperature of the topping cycle. The maximum temperature must be lower than the exhaust temperature by some temperature difference, which was chosen as 20° C for this project. Finally, the minimum temperature (compressor inlet temperature) is limited to 36.85° C.

### **Pressure Losses**

In the idealized Brayton cycle, all heating and cooling occurs isobarically, meaning that the pressure remains constant. Supercritical CO<sub>2</sub> has a low viscosity, but nevertheless, pressure losses in the flow still occur within the heat exchangers and recuperators, and need to be accounted for while modeling evaluating the performance of a power cycle. In all of the following analyses, a 0.5% loss in pressure was assumed through each heat exchanger and through each recuperator.

### **Isentropic Efficiencies of Turbines and Compressors**

For the idealized Brayton cycle, the working fluid is compressed and expanded isentropically, meaning that the entropy of the working fluid remains constant throughout the process. This will only occur if no heat is transferred to or from the working fluid during the process, and if the process is completely reversible. Because of inherent irreversibilities in actual processes, no real expansion or compression process is truly isentropic, so an assumed isentropic efficiency is used to describe the relative performance of the compressors and turbines. This analysis uses an assumed 85% isentropic compressor efficiency and an assumed 90% isentropic turbine efficiency.

## **Recuperator and Heat Exchanger Constraints**

A recuperator/heat exchanger can be constrained in one of two ways. The effectiveness may be defined, which describes the quantity of heat transferred in relation to the maximum possible quantity of heat transferred. The other way to constrain the system is by defining the minimum terminal temperature difference, or TTD. In the interest of maintaining control over the turbine inlet temperature(s), the TTD method was chosen. A low TTD corresponds to a high recuperator effectiveness, and a high recuperator effectiveness generally corresponds to a more efficient power-cycle. A conflict arises, however, because the cost of a highly effective recuperator is a much greater than the cost of a less effective recuperator. To keep costs within reason, all TTDs were chosen to be 20° C in this report.

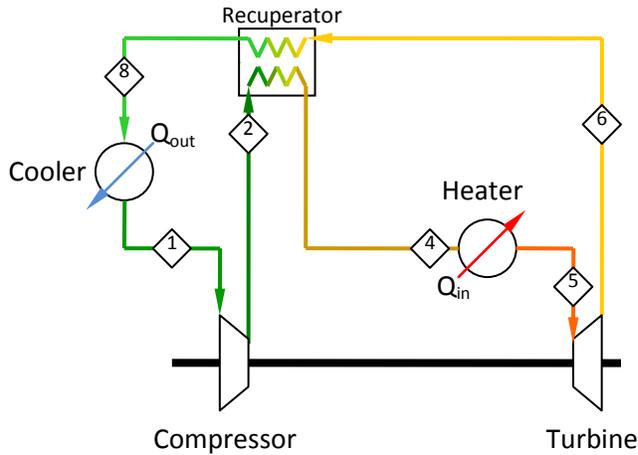
## **Additional Assumptions**

The system is assumed to be in steady-state operation. The following analysis does not include information regarding performance while the cycle is ramping up or ramping down. The system is assumed to be adiabatic, with no heat being lost to the surroundings, other than from the precooler or intercooler if applicable.

# **Initial Modeling and Analysis**

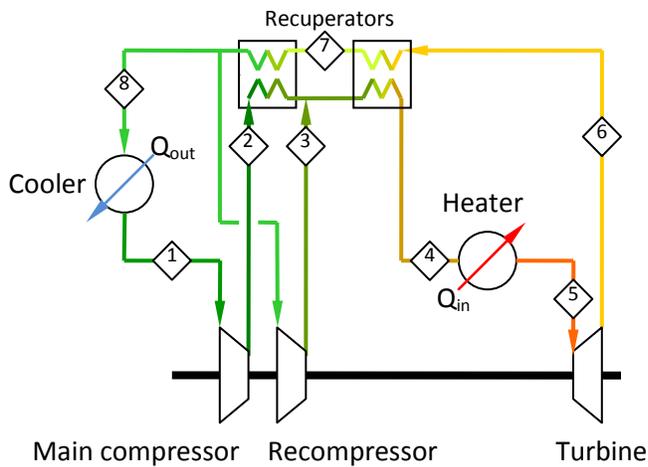
At the start of this project, three proposed bottoming cycles were modeled, using the maximum exhaust temperature of the Siemens H class minus the appropriate TTD (20°C) as the maximum cycle temperature. All modeling efforts utilized MATLAB software, which accessed the NIST Reference Fluid Thermodynamic and Transport Properties Database (REFPROP) for

all thermodynamic properties, and made the necessary calculations in order to predict the operating conditions and performance of each sCO<sub>2</sub> cycle. Shown below are the three cycles modeled initially, along with their optimized thermal efficiency.



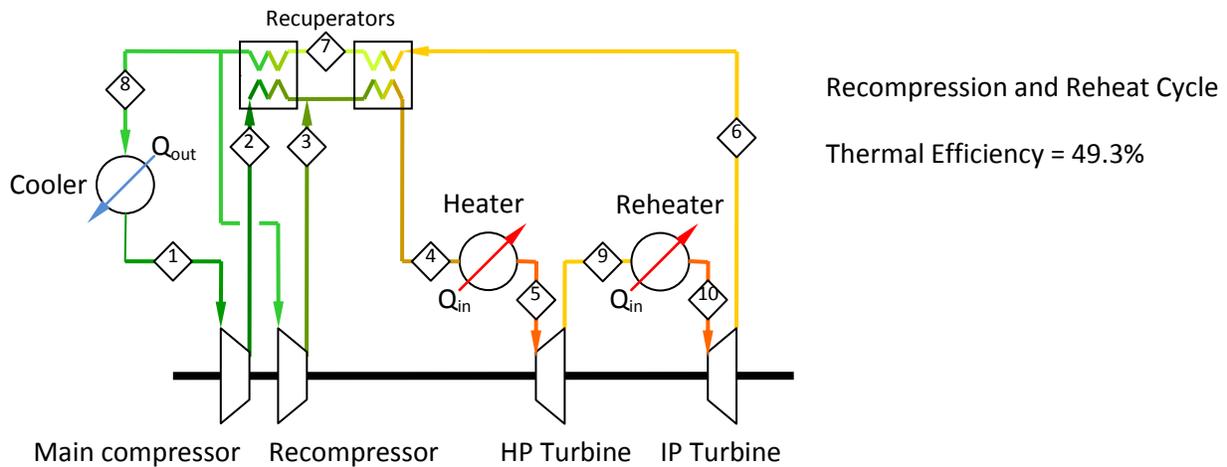
Simple Recuperated Cycle  
Thermal Efficiency = 40.3%

**Figure 2. Simple Recuperated Cycle Flowsheet**



Recuperated Cycle with Recompression  
Thermal Efficiency = 45.4%

**Figure 3. Recuperated Recompression Cycle Flowsheet**



**Figure 4.     Recompression and Reheat Cycle Flowsheet**

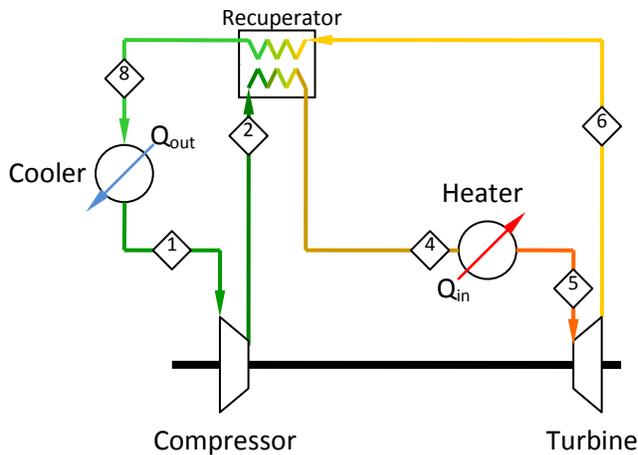
## Sensible Heat vs. Heat Flux Sources

After modeling the three proposed cycles, the results seemed to agree with claims that an sCO<sub>2</sub> Brayton cycle may have slightly higher thermal efficiency than a Rankine steam cycle operating under the same maximum and minimum temperature. However, new information soon came into light, and additional considerations had to be taken into account.

A presentation by Echogen Power Systems included a discussion regarding the merits of a high efficiency power cycle. According to the presentation, a high cycle efficiency does not necessarily mean that it is well suited to a bottoming cycle application.<sup>[5]</sup> A distinction needs to be made between two different heat sources that are commonly encountered within the power generation industry. A heat flux source is one in which heat is not limited by temperature reduction. A good example of a heat flux source is a nuclear reaction. A nuclear power source will produce a constant heat flux regardless of the temperature in which the heat is being transferred. In contrast, a *sensible* heat source is one in which the heat available is heavily dependent on the temperature. A good example of a sensible heat source is the exhaust from a

combustion cycle. The heat available from a sensible heat source is roughly proportional to its temperature change, that is,  $Q \approx m_{\text{exhaust}} * c_{P,\text{avg}} * \Delta T_{\text{exhaust}}$ . A consequence of this property is that only a small portion of the available waste heat is recoverable at high temperatures. The commonly proposed cycles shown above are well suited to operating with a heat flux producing power source, but are not well suited to a sensible heat source, such as topping cycle exhaust.

For a more specific example of this effect, consider the simple recuperated cycle shown in Figure 5.



**Figure 5. Simple Recuperated Cycle Flowsheet**

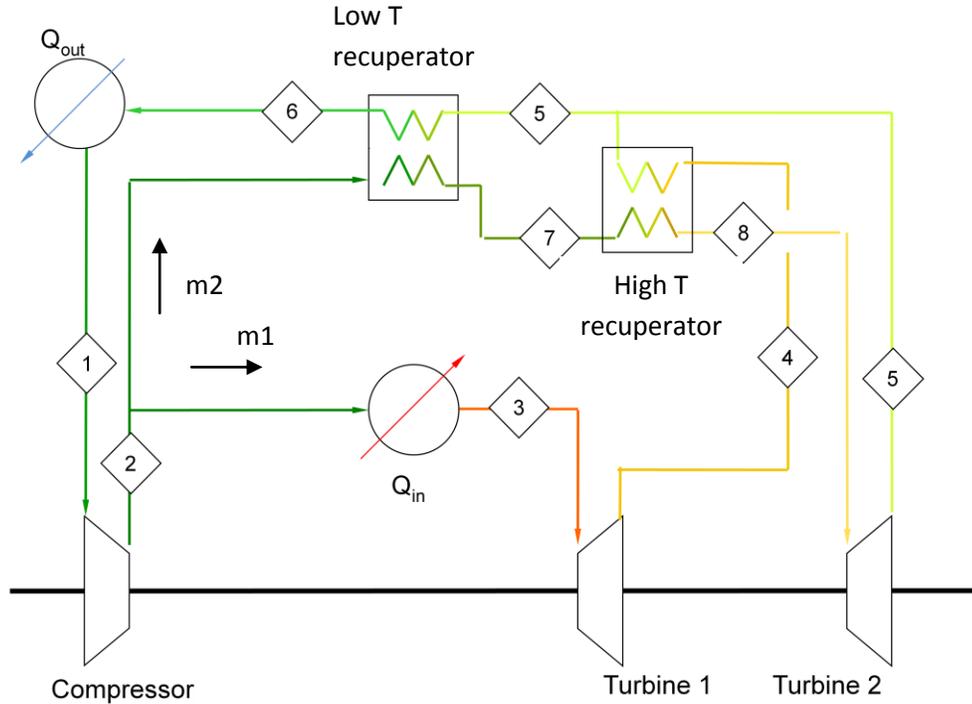
While this cycle theoretically exceeds 40% efficiency when operating in tandem with the H class, the heater shown needs to be considered in greater detail. It is, in fact, a heat exchanger, whose goal is to transfer heat from the hot exhaust gas (its heat source), to the sCO<sub>2</sub>. For a heat exchanger to function, a temperature gradient must exist between the two flows, that is to say, the exhaust will always be at a higher temperature than the sCO<sub>2</sub>. This means that the exhaust

gases will only supply heat until it reaches a temperature marginally greater than  $T_4$ . Due to the recuperator,  $T_4$  is considerably higher than the ambient temperature, in the vicinity of  $400^\circ\text{C}$ , which means that the exhaust gases, which are this cycle's heat source, will only be cooled to perhaps  $420^\circ\text{C}$ . This  $420^\circ\text{C}$  exhaust can no longer supply any heat to the system, so it exits, carrying with it a significant portion of its initial thermal energy. From its initial temperature of  $625^\circ\text{C}$ , the exhaust has only been cooled by  $205^\circ\text{C}$ . If instead this exhaust were cooled to say,  $215^\circ\text{C}$ , the temperature change has doubled, which means that the heat input to the system has roughly doubled.

To put it another way, a bottoming cycle may have a thermal efficiency of 50%, but if only 100 MW of heat is supplied to the system, the cycle will only produce 50 MW of usable power. On the other hand, another power cycle might have a thermal efficiency of only 30%, but if it can recover 300 MW of heat from the exhaust, it will produce 90 MW of usable power, nearly double the power of the first cycle. So, while thermal efficiency is certainly an important quality of any power cycle, a high cycle efficiency does not necessarily correlate with a high power output when used with a sensible heat source. A truly effective bottoming cycle would be one in which thermal efficiency is balanced with the ability to recover waste heat. For this reason, in the remainder of this report, the power output of each cycle will be used as the primary measure of each cycle's performance.

# Baseline Cascade Cycle (Cycle 1) Modeling

Included in the Echogen presentation was a simple, baseline idea for an sCO<sub>2</sub> bottoming cycle.



**Figure 6. Cycle 1 Flowsheet**

The cycle consists of one compressor that supplies all of the high pressure, low temperature CO<sub>2</sub>. The compressor outlet stream is split into two separate flows. The first stream travels through the heat exchanger, recovering heat from the topping cycle exhaust. This first stream is expanded in turbine 1, and is then used to heat the other stream, which is expanded in turbine 2. The advantage of this configuration is that it recovers a large portion of the waste heat from the topping cycle. Due to the absence of a recuperator on the first stream, the exhaust reaches a temperature marginally above  $T_2$ , which is far lower than  $T_4$  from the previous case. A

much larger portion of the topping cycle waste heat is recovered, at the cost of thermal efficiency. Shown below are the optimized cycle conditions for the H Class and LM6000 cases.

Also shown is the shaft power and efficiency of each cycle.

**Table 2. Cycle 1 on H Class, state properties**

State	1	2	3	4	5	6	7	8
Pressure(Bar)	85.00	276.00	274.62	85.86	85.43	85.43	274.62	273.25
Temperature (°C)	36.85	86.37	604.85	461.25	313.72	173.67	292.73	441.25
Enthalpy(kJ/kg)	332.12	367.96	1099.63	938.23	767.19	607.23	704.70	893.76
Entropy(kJ/kg*K)	1.4258	1.4408	2.7543	2.7790	2.5201	2.2085	2.1980	2.4958

$m_1 = 52.5\%$   $m_2 = 47.5\%$

Power Output=133 MW

Efficiency = 28.38%

**Table 3. Cycle 1 on LM6000, state properties**

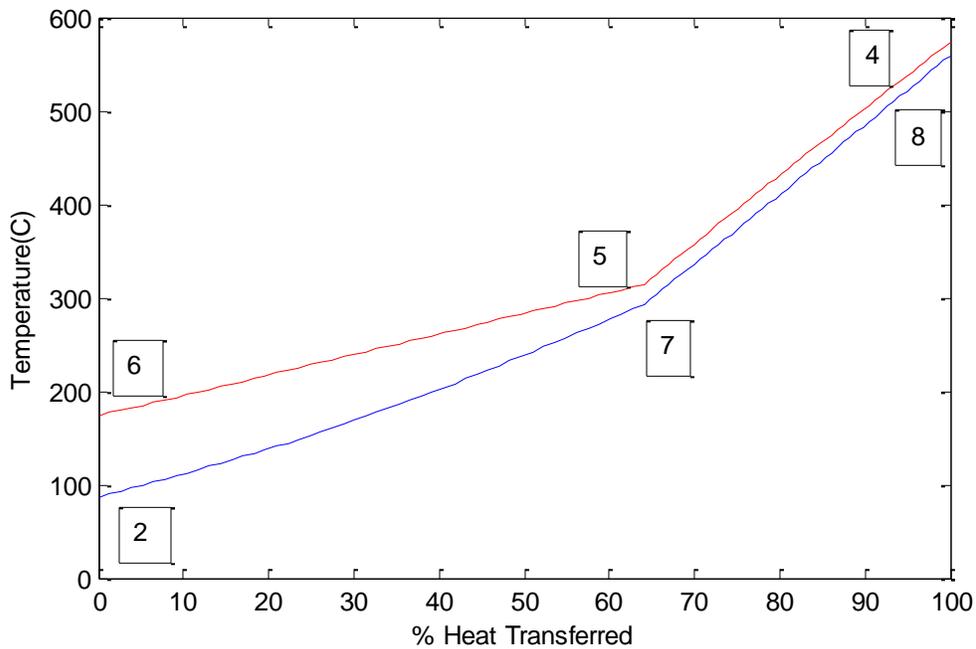
State	1	2	3	4	5	6	7	8
Pressure(Bar)	85.00	276.00	274.62	85.86	85.43	85.43	274.62	273.25
Temperature (°C)	36.85	86.37	450.85	322.38	187.64	129.01	167.51	302.38
Enthalpy(kJ/kg)	332.12	367.96	905.68	776.99	623.45	553.61	525.26	717.53
Entropy(kJ/kg*K)	1.4258	1.4408	2.5115	2.5357	2.2443	2.0820	1.8378	2.2214

$m_1 = 55.6\%$   $m_2 = 44.4\%$

Power Output = 14.2 MW

Efficiency = 25.92%

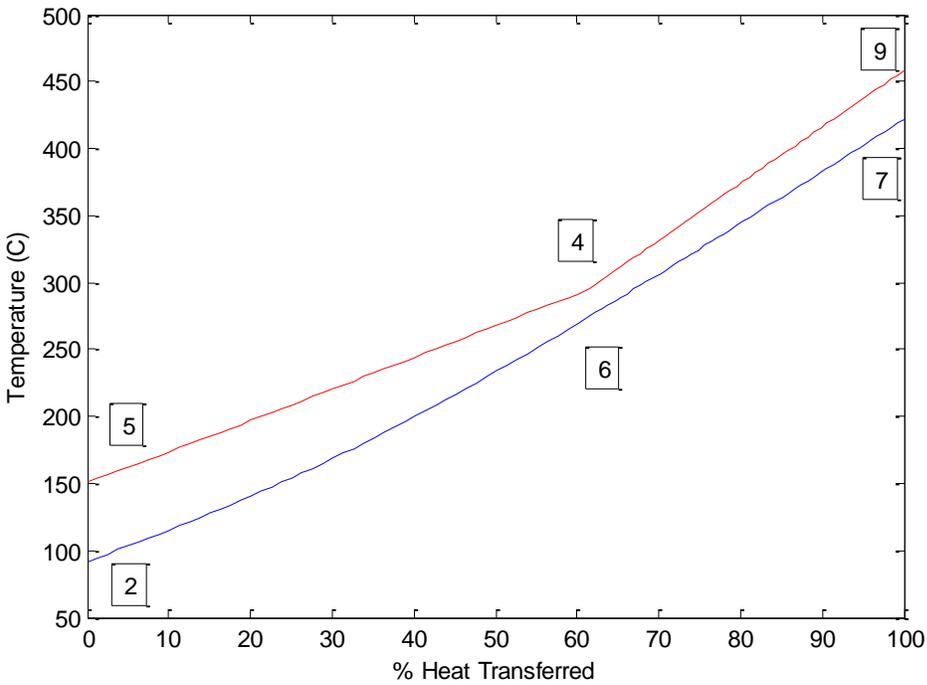
While this cycle displays good heat recovery, a major drawback of the design is that the low temperature recuperator is not very effective, due to a higher mass flow rate on the hot side than on the cold side. The cold side of the recuperator contains only the sCO<sub>2</sub> that is going toward turbine 2, while the hot side contains the recombined outlet streams of both turbines. Shown below is an example temperature profile of the two recuperators in Cycle 1.



**Figure 7** Temperature profile of Cycle 1 recuperators with labeled states

The temperature at state 6 is significantly higher than the temperature at state 2. The flow at state 6 goes straight to the pre-cooler, which rejects this thermal energy to a heat sink, rather than being used to heat the high pressure sCO<sub>2</sub> on the way to a turbine. Ideally, T<sub>6</sub> would reach a





**Figure 9** Temperature profile of Cycle 2 recuperators

In this example temperature profile for cycle 2, one can see that the precooler inlet temperature is now approximately 151° C, down from 174° C in the first cycle. This extra recuperated heat leads to a slightly higher predicted shaft power than the first case. The state properties of this cycle when used on the H Class and LM6000 are shown with the associated efficiency and shaft power below.

**Table 4.** Cycle 2 on H Class, state properties

State	1	2	3	4	5	6	7	8	9
Pressure(Bar)	84.00	276.0	274.6	84.85	84.42	277.4	278.8	280.2	85.27
Temperature (°C)	36.85	91.76	421.85	295.0	150.7	274.5	421.9	604.85	458.1
Enthalpy(kJ/kg)	341.6	379.5	869.27	745.9	580.5	680.1	868.8	1099.3	934.6

Entropy(kJ/kg*K)	1.457	1.472	2.4601	2.489	2.149	2.152	2.457	2.750	2.775
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m1=45% m2=55%

Power = 145 MW

Efficiency = 31.18%

**Table 5. Cycle 2 on LM6000, state properties**

State	1	2	3	4	5	6	7	8	9
Pressure(Bar)	84.00	276.0	274.6	84.85	84.42	277.4	278.8	280.2	85.27
Temperature (°C)	36.85	91.76	246.9	136.5	117.6	116.8	246.9	451.2	319.8
Enthalpy(J/kg)	341.6	379.5	643.3	563.2	539.8	431.5	642.5	905.6	774.2
Entropy(J/kg*K)	1.457	1.473	2.085	2.107	2.049	1.610	2.081	2.507	2.532

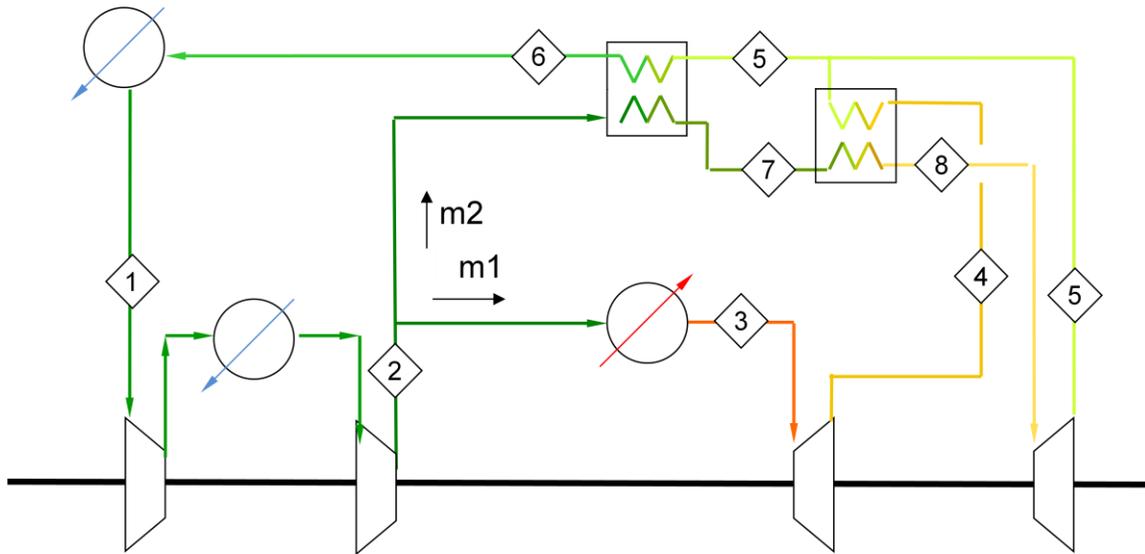
m1=55% m2=45%

Efficiency = 24.78%

Power = 13.4 MW

## Effect of Intercooling on Cycle Performance

Adding an intercooler to the compression process was found to increase the net power output of the previous two cycles. The intercooler decreases the total compression work, and it also decreases the minimum temperature of the high pressure sCO<sub>2</sub>. The lower temperature at the compressor outlet increases the heat transfer in the low temperature recuperator, and also cools the topping cycle exhaust gases to a lower temperature, improving the ability of the cycle to recover heat from the topping cycle exhaust gases.



**Figure 10. Cycle 1 with intercooler flowsheet**

**Table 6. Intercooled Cycle 1 on H Class, state properties**

State	1	2	3	4	5	6	7	8	9	10	11
Pressure(Bar)	60.00	89.00	88.56	276.0	274.6	60.60	60.30	274.6	273.3	271.9	60.91
Temperature (°C)	36.85	69.72	36.85	76.62	370.9	218.1	102.9	197.7	370.8	604.8	424.7
Enthalpy(kJ/kg)	445.7	464.6	314.9	347.1	805.0	666.6	537.9	572.9	805.2	1099.	898.8
Entropy(kJ/kg*K)	1.8270	1.8353	1.3681	1.3820	2.3641	2.3958	2.0976	1.9425	2.3653	2.7564	2.789

$m_1 = 43\%$   $m_2 = 57\%$

Power = 161 MW

Efficiency = 33.71 %

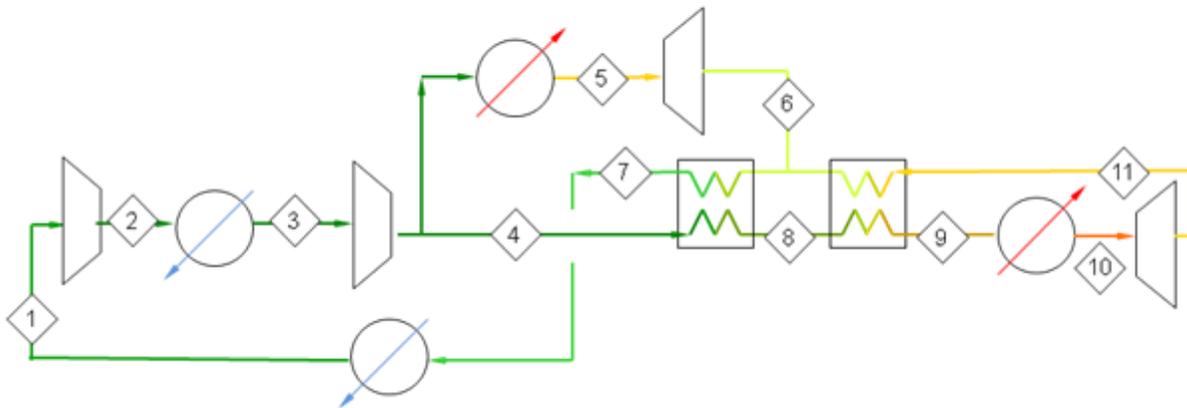
**Table 7. Intercooled Cycle 1 on LM6000, state properties**

State	1	2	3	4	5	6	7	8	9	10
Pressure(Bar)	50.00	89.00	88.56	276.0	274.6	50.76	50.50	50.25	274.6	273.3
Temperature (°C)	36.85	86.90	36.85	76.62	450.9	272.0	99.17	98.12	77.75	252.0
Enthalpy(J/kg)	462.4	493.6	314.8	347.1	905.7	728.6	540.3	539.3	349.7	650.7
Entropy(J/kg*K)	1.905	1.918	1.368	1.3820	2.5115	2.5481	2.1331	2.1312	1.390	2.099

$m_1 = 61.5\%$     $m_2 = 38.5\%$

Power = 14.4 MW

Efficiency = 25.60%



**Figure 11. Cycle 2 with intercooler flowsheet**

**Table 8. Intercooled Cycle 2 on H class, state properties**

State	1	2	3	4	5	6	7	8	9	10	11
Pressure(Bar)	60.00	89.00	88.56	276.0	274.6	60.60	60.30	274.6	273.3	271.9	60.91

Temperature (°C)	36.85	69.72	36.85	76.62	370.8	218.1	102.9	197.7	370.9	604.9	424.7
Enthalpy(kJ/kg)	445.7	464.6	314.9	347.1	805.0	666.6	537.9	572.9	805.1	1100.	898.8
Entropy (kJ/kg*K)	1.827	1.835	1.368	1.382	2.364	2.396	2.098	1.943	2.365	2.756	2.789

$m_1 = 43\%$   $m_2 = 57\%$

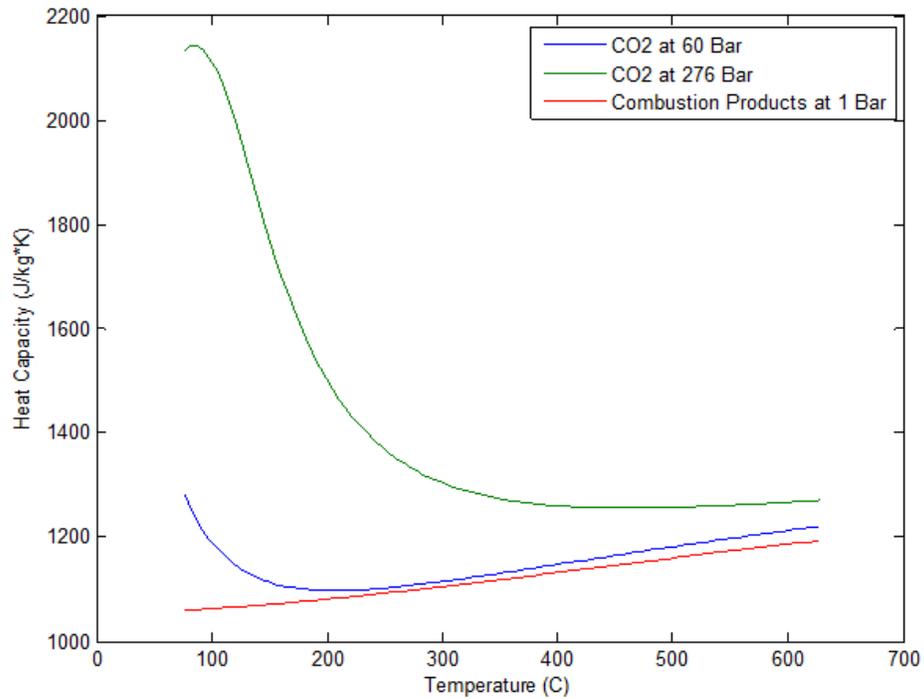
Power = 161 MW

Efficiency = 33.71 %

The addition of an intercooler into these cycles increases the net power significantly. The power output of Cycle 1 increased by 12% on the H Class and 1.4% on the LM-6000. The power output of Cycle 2 increased by 11% and 10% on the H Class and LM-6000 respectively. This data suggests that the use of an intercooler should be considered for any sCO<sub>2</sub> cycle.

## Cycle 3

On Cycle 1 and Cycle 2, the recuperator temperature profiles still did not track very well, so I decided to use a different approach for the last cycle. By studying a graph of the specific heat capacity of high pressure sCO<sub>2</sub>, low pressure sCO<sub>2</sub>, and exhaust gas, I noticed that high pressure sCO<sub>2</sub> has a very large heat capacity at lower temperatures, which drops off as the temperature rises. The low pressure sCO<sub>2</sub> also has a higher heat capacity at low temperatures, but it does not change as drastically.



**Figure 12. Specific Heat Capacity vs Temperature**

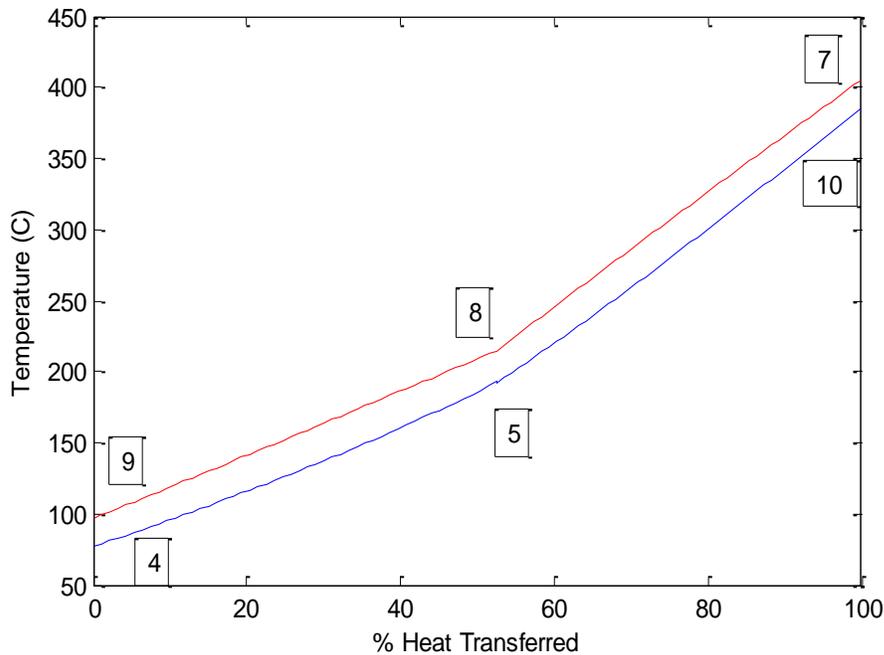
Using this graph, I designed a new cycle that takes advantage of this varying heat capacity by controlling the sCO2 flow rates through the heat exchangers to maximize heat transfer.



with the high heat capacity of the high pressure, low temperature stream allows the temperature profile of the low T recuperator profile to track very closely.

The flows are recombined at state 5, then split again, this time sending a higher ratio of the sCO<sub>2</sub> to be heated by the exhaust gases. By finding the optimal mass flow rates, the temperature profiles of both recuperators can be more precisely controlled, ultimately leading to a larger quantity of recuperated heat, without compromising the cycle's ability to recover waste heat from the topping cycle.

Shown is an example temperature profile of Cycle 3



**Figure 14. Temperature profile for Cycle 3 recuperators**

As can be seen, the temperature profile of both recuperators track much more closely than in Cycles 1 or 2, and the precooler inlet temperature approaches the minimum temperature

difference from the compressor outlet temperature, signifying that the quantity of recuperated heat cannot be increased much further.

**Table 9. Cycle 3 on H Class, state properties**

State	1	2	3	4	5	6	7	8	9	10
Temp(°C)	36.85	70.69	36.85	75.08	232.28	604.85	424.14	247.92	90.98	404.10
Pressure(bar)	50.00	89.00	88.56	276.00	274.62	273.25	50.76	50.50	50.25	27324.69
Enthalpy(kJ/kg)	462.41	492.22	314.87	345.64	565.03	1099.69	877.53	666.54	528.32	822.91
Entropy(kJ/kg*K)	1.9049	1.9141	1.3681	1.3778	1.9257	2.7554	2.7924	2.4287	2.1014	2.3925

m1 = 37%    m2 = 63%

m3 = 55%    m4 = 45%

Power = 169 MW

Efficiency = 35.20%

**Table 10. Cycle 3 on LM6000, state properties**

<b>State</b>	<b>1</b>	<b>2</b>	<b>3</b>	<b>4</b>	<b>5</b>	<b>6</b>	<b>7</b>	<b>8</b>	<b>9</b>	<b>10</b>
Pressure(Bar)	60.00	89.00	88.56	276.0	274.6	273.3	60.91	60.60	60.30	273.3
Temperature (°C)	36.85	69.72	36.85	76.62	105.0	451.1	289.7	129.1	98.55	269.7
Enthalpy(kJ/kg)	445.7	464.6	314.9	347.1	407.9	906.1	745.7	567.9	532.7	674.5
Entropy(kJ/kg*K)	1.827	1.835	1.368	1.382	1.550	2.513	2.545	2.174	2.084	2.144

m1 = 42%    m2 = 58%

m3 = 60%    m4 = 40%

Power = 15.2 MW

Efficiency = 27%

# Summary of Results

The resultant power outputs of each sCO<sub>2</sub> cycle when used with either the Siemens H Class topping cycle or LM6000 topping cycle are tabulated in Table 11, along with the benchmark steam bottoming cycle, found in bold at the bottom of the table.

**Table 11. Summary of Power Output for all Cycles**

	<b>H Class</b>	<b>LM6000</b>
Cycle 1 Power	133 MW	14.2 MW
Intercooled Cycle 1 Power	149 MW	14.4 MW
Cycle 2 Power	145 MW	13.4 MW
Intercooled Cycle 2 Power	161 MW	14.8 MW
Intercooled Cycle 3 Power	169 MW	15.2 MW
<b>Current Steam Bottoming Cycle Power</b>	<b>195 MW</b>	<b>14.0 MW</b>

For the H Class application, none of the proposed cycles meet the power output of the steam cycle currently in use. In contrast, all of the sCO<sub>2</sub> cycles have a higher output than the steam cycle when used on the LM6000, save for the non-intercooled Cycle 2. The use of an intercooler in each cycle increases the power output of the cycle for both the H Class and LM6000 application. In both the LM6000 and H Class cases, Cycle 3 produces the highest power output.

## Conclusions and Recommendations

On the LM6000, the top performing cycle, Cycle 3, yields a theoretical power output that is 8.6% higher than the benchmarked steam bottoming cycle. On the other hand, none of the cycles modeled for the high temperature, high power H Class application exceeded the power output of its current steam cycle. Taking only power output into consideration, these findings suggest that an sCO<sub>2</sub> bottoming cycle may be better suited to low temperature applications.

A more thorough analysis of these configurations, however, should also include a cost-benefit analysis, in order to better gauge the advantages and disadvantages of an sCO<sub>2</sub> cycle in comparison to the standard Rankine steam cycle. While the highest performing sCO<sub>2</sub> cycle prediction yields only 86.7% of the power output of the current bottoming cycle in place on the H Class, it may still be an attractive alternative, depending on the capital and maintenance costs associated with its implementation. Cycle 3 is recommended for further research, as well as a cost-benefit analysis regarding its potential use as a bottoming cycle for high power, high exhaust temperature gas turbines.

The choice of recommended cycle configuration is not as straightforward for a low power, low temperature application. In the high power case, the associated complexity and cost of Cycle 3 in comparison to Cycles 1 and 2 is probably justified for the increased power output. The comparison is not so simple in the other application. The difference in power between Cycle 1 and Cycle 3 is on the order of 1 MW, which may not be enough to justify the added complexity of Cycle 3 in comparison to Cycle 1. All 3 cycles are recommended for further research and a cost-benefit analysis with regard to its potential use as a bottoming cycle in low power, low temperature applications.

Anyone involved in researching or developing an sCO<sub>2</sub> bottoming cycle should be wary of using cycle efficiency to describe the performance of a cycle configuration. Bottoming cycle performance should be compared in the context of power output, rather than efficiency, because bottoming cycles operate using a sensible heat source. The ideal sCO<sub>2</sub> bottoming cycle will correctly balance the importance of a high thermal efficiency with the ability of the cycle to recover the maximum amount of heat from its source.

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