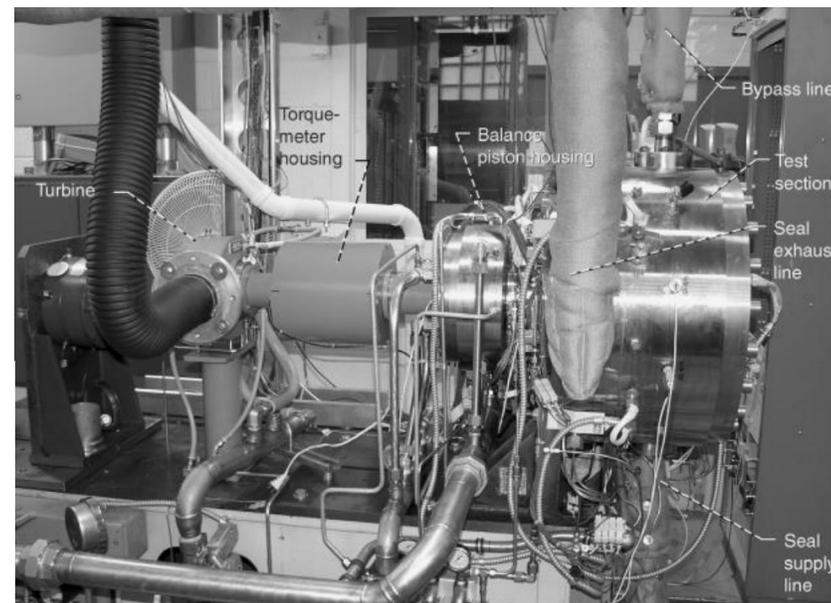


# High-Temperature Dry Gas Seal (HTDGS) Test Rig

**University Turbine Systems Research (UTSR)**  
**Gas Turbine Industrial Fellowship Program**  
Contributions from: Elliot Moore

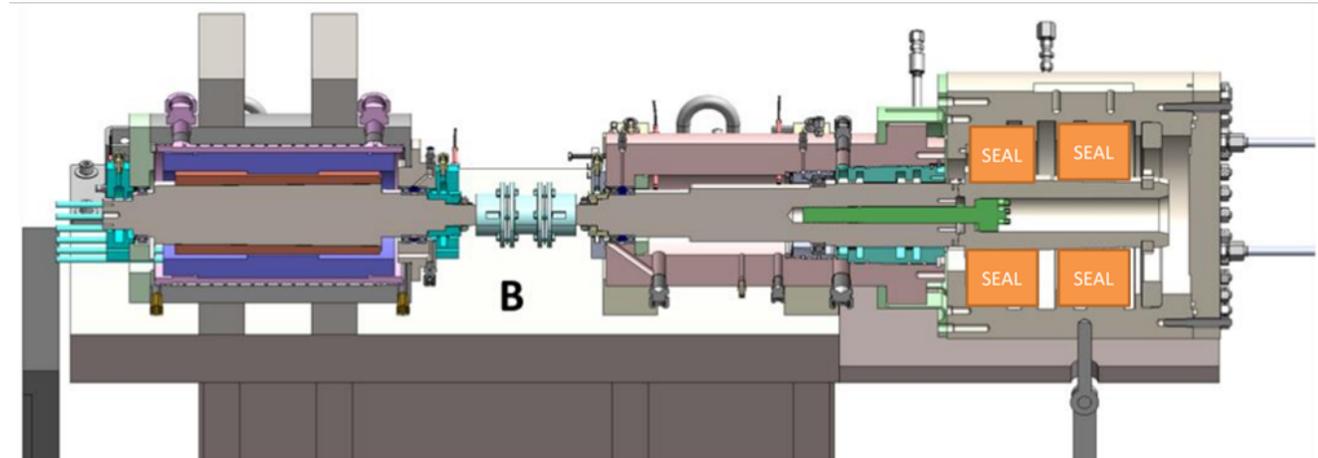
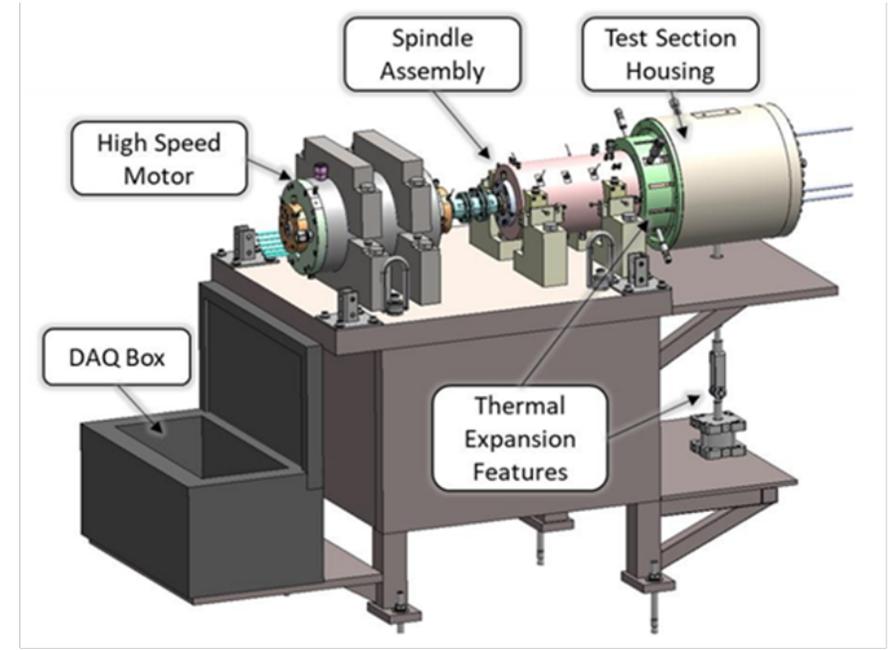
# Background

- **Objective:**
  - Test rig capable of testing Dry Gas Seals (DGS) at both high temperatures and pressures (700C, 250 bara)
- **Previous Work:**
  - High temperature; low pressure
    - NASA test rig (650C, 5 bara)
    - Tested finger seals
  - High pressure; low temperature
    - Vannini (500 bar, ambient temp)
    - Tested labyrinth seals

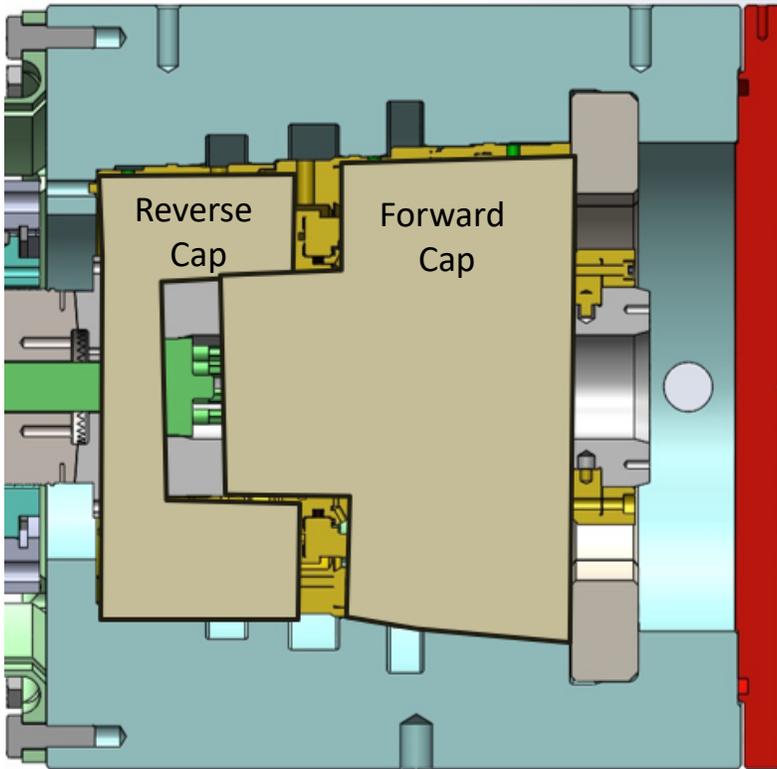


# Test Rig

- **Goal:**
  - DGS test at 700C and 250 bara
  - Versatility with various seal testing
- **General Layout:**
  - High speed motor assembly:
    - Operating at 21 krpm and 250 kW
  - Spindle assembly:
    - Ground point, thermal management system
  - Test housing assembly:
    - Hirth coupling to accommodate various seal designs
    - Pressure management system
- **Design Challenges:**
  - Thermal growth management
  - Bearing Temperatures
  - **Pressure management**



# Hydro Testing



## Objective:

- Develop scheme to test pressure containment of seal housing according to Boiler Pressure Vessel Code (BPVC)
- Design plug caps to simulate operation pressure under hydrostatic loading
  - Dimension according to desired running clearances

## Contributions:

- Dimensioned parts and sent for machining

# End Cap Pressure Analysis

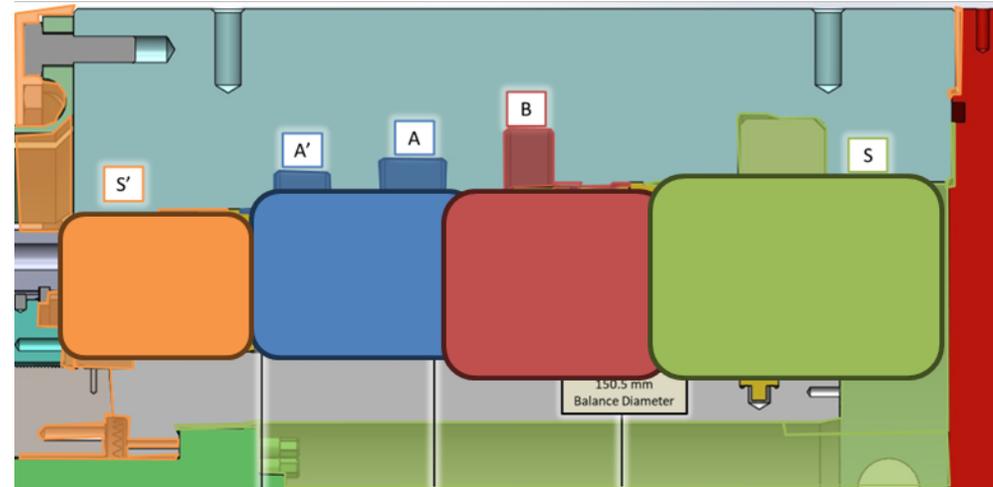
## Problem:

- S chamber:
  - Normal Operation: **~1.52 bara**
  - MAWP (Hi-Temp): **~2.5 bara**
  - MAWP (Low-Temp): **~2.67 bara**
- Rapid pressure increase due to high  $\Delta P$  (B→S)
- Unknown pressure build-up as pressure is relieved

## Objective:

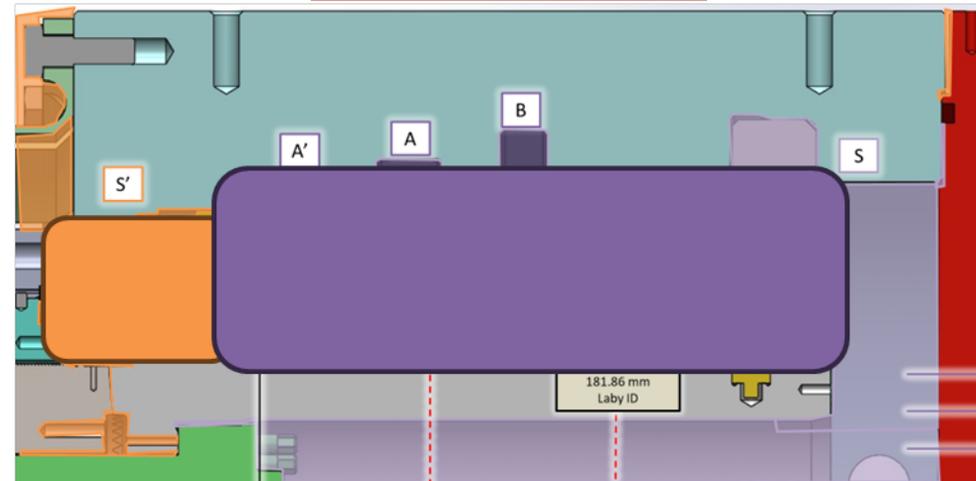
- Determine back pressure build-up upon tandem seal failure
- Develop successful pressure relieving strategy (Max Press. < MAWP)
  - Burst disk utilized on end cap

## Normal Operation



<b>MAWP-Hi</b>	
<b>A Zone:</b>	
• Max Temp:	700C
• Max Press:	250bara
<b>B Zone:</b>	
• Max Temp:	700C
• Max Press:	10.0bara
<b>S Zone:</b>	
• Max Temp:	700C
• Max Press:	1.52bara
<b>S' Zone:</b>	
• Temp:	Adiabatic
• Max Press:	Ambient

## Tandem Seal Failure



<b>Tandem Failure -Hi</b>	
<b>A Zone:</b>	
• Max Temp:	700C
• Max Press:	66.2bar
<b>B Zone:</b>	
• Max Temp:	700C
• Max Press:	66.2bar
<b>S Zone:</b>	
• Max Temp:	700C
• Max Press:	<b>2.5bara</b>
<b>S' Zone:</b>	
• Temp:	Adiabatic
• Max Press:	Ambient

**MAWP prev. determined**

# End Cap Pressure Analysis – Model

**Objective:** Model ruptured burst disk as *orifice* flow model and determine pressure in the S chamber

**Mass Balance Governing Equations:**

$$\frac{dM_{A+B}}{dt} = \dot{m}_{in,ext} - \dot{m}_{B \rightarrow S} \quad \frac{dM_S}{dt} = \dot{m}_{B \rightarrow S} - \dot{m}_{out}$$

**Flow in:**

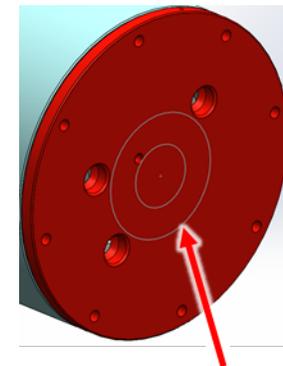
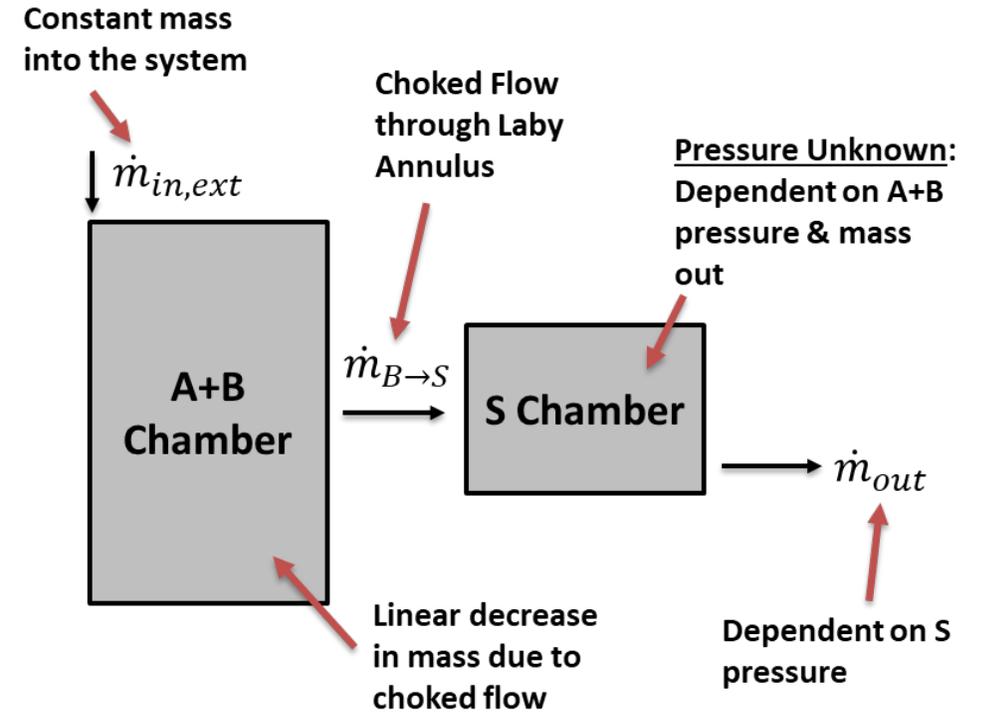
$\dot{m}_{in,ext}$ : Constant CO<sub>2</sub> flow into system

**Flow A+B → S**

$\dot{m}_{B \rightarrow S} = \rho_{A+B} \cdot v_{SoS} \cdot A_{laby}$ : Choked flow through laby annulus

**Flow S → ambient**

$\dot{m}_{out} = C_d \cdot \varepsilon \cdot A_{orif} \sqrt{\frac{2\rho_S \cdot (P_S - P_{atm})}{1 - \beta^4}}$ : Flow through orifice (burst disk)



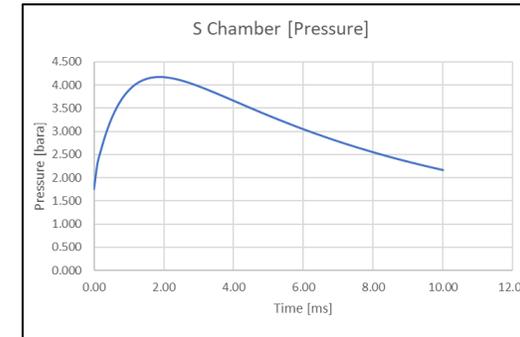
Proposed Burst Disk Location [orifice model]

# End Cap Pressure Analysis – Initial Results

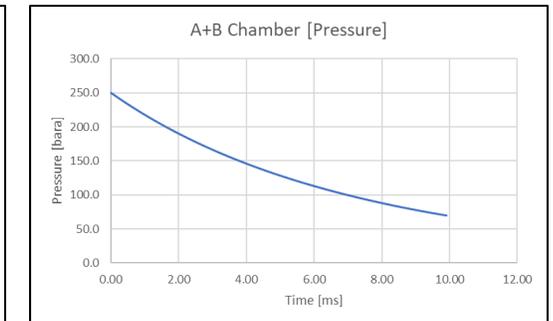
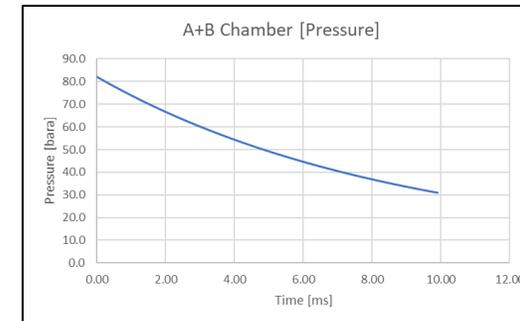
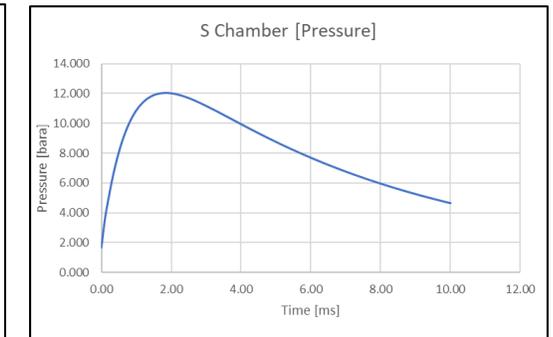
Burst Disk Diameter: 6 in			
Test	Parameter	Value	Unit
Low-Test Point [500C]	A+B Operating Pressure	82	bara
	S Operating Pressure	1.52	bara
	S Burst Pressure	1.747	bara
	Max S Pressure	4.174	bara
Hi-Test Point [700C]	A+B Operating Pressure	250	bara
	S Operating Pressure	1.52	bara
	S Burst Pressure	1.747	bara
	Max S Pressure	12.065	bara

*Remember... [MAWP]*  
 Hi: ~2.5 bara  
 Low: ~2.67 bara

## Low-Test



## Hi-Test



**High S chamber pressures results in significant forces on End Cap**

### Limiting Factors:

- Large laby seal clearance (40 mils) → high mass flow to S chamber
- High pressure ratio A+B → S
- Burst Disk diameter restriction

### Proposed Solutions:

- End cap re-design to accommodate increased load
- Incorporate larger burst disk (> 6in)
  - Cons: Increasingly expensive as diameter increases; even more if custom design is required
- Incorporate PSV pressure relief before S chamber reaches maximum pressure
  - Add PSV to relieve pressure in B chamber before secondary seal failure

# End Cap Pressure Analysis – Revised Results

Thus far, model assumes tandem seal failure ∴ high pressure region rapidly builds pressure in S chamber

### Implementing PSV into Chamber B:

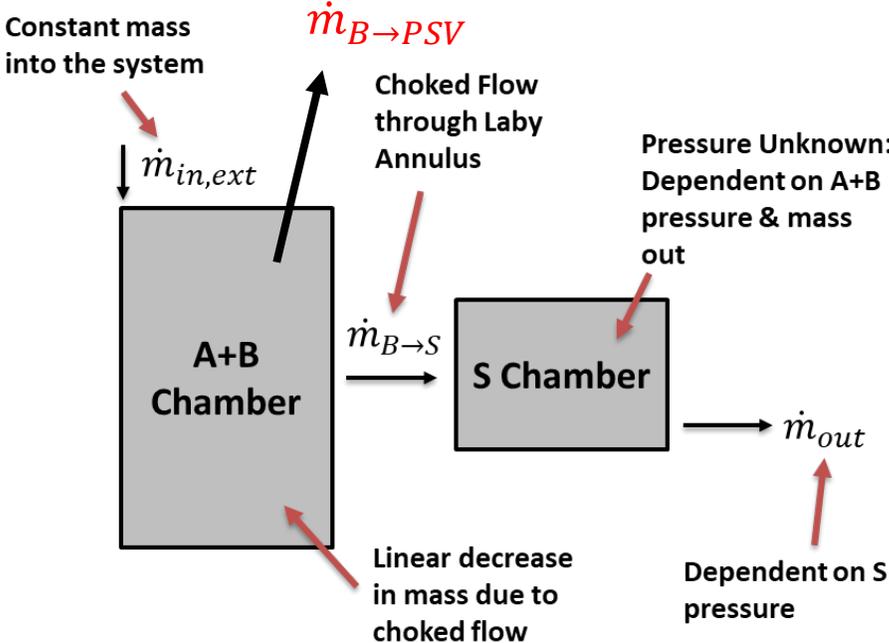
- Incorporate PSV pressure relief from B chamber (nominally 10 bara)
  - After primary seal failure (A→B), B chamber will build pressure above 10 bara.
  - PSV engages and relieves pressure **before secondary failure**

### Incorporation into model:

- PSV set at specified pressure (estimate 20 bara for model purposes)
- Assume PSV sustains specified pressure in chamber B as pressure is relieved

Burst Disk Diameter: 4* in			
Test	Parameter	Value	Unit
Low-Test Point [500C]	A+B Seal Failure Pressure	20	bara
	S Operating Pressure	1.52	bara
	S Burst Pressure	2.137	bara
	Max S Pressure	2.485	bara
Hi-Test Point [700C]	A+B Seal Failure Pressure	20	bara
	S Operating Pressure	1.52	bara
	S Burst Pressure	2.137*	bara
	Max S Pressure	2.487	bara

*\*Note: 4 in burst disk modeled in analysis rather than 6 in*



*PSV reduces force on end cap and increases control in pressure relief system*

### Future Work Recommendations:

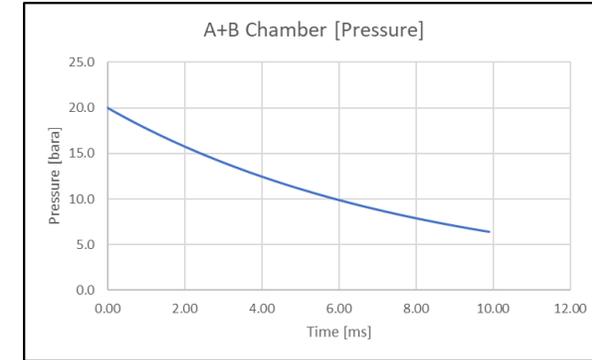
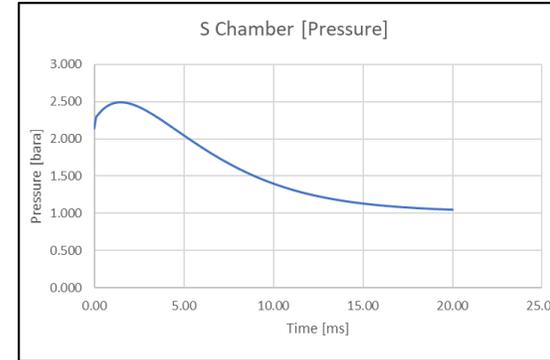
- Implement PSV into test rig P&ID
- Determine and incorporate factor of safety into PSV actuation pressure

# End Cap Pressure Analysis – Modeling Tool

## Pressure Analysis Excel Tool:

- Input/ determined via 'Goal Seek'
- Input parameters

Ambient Conditions				A+B to S Chamber Initial Conditions			
Description	0	0	Notes	Parameters	Values	Units	Notes
T	700.0	C	Isothermic Temp	P1	20.0	bara	A+B Cavity Pressure
P <sub>3</sub>	1.000	bara	Ambient Pressure		2.00E+03	kPa	
	100.000	kPa			2.00E+06	Pa	
	1.00E+05	Pa		v1	468.70	m/s	Speed of sound (choke flow)
				V1	0.00218	m <sup>3</sup>	Volume of A+B Chamber/ CAD
				area1	5.75E-04	m <sup>2</sup>	Laby annulus
				Z <sub>1,0</sub>	1.0039	-	Initial Comp. Factor/ REFPROP
				Min,ext	0.1159	kg/s	Const./ mass flow into system
					0.023614		
				M <sub>0</sub>	7	kg	Initial mass
Initial Conditions of S Chamber				Laby Seal			
Parameters	Values	Units	Notes		39.37	mils	Calc
C <sub>d</sub>	0.600	-	Burst Disk Cd estimation	Laby Clearance	1	mm	
P <sub>2,0</sub>	1.162	kg/m <sup>3</sup>	S Chamber initial density		1.00E-03	m	
d2	9.9213	in	Upstream diameter	Laby D(ID)	182	mm	HDGS1130-TA1-U-Final (EB Seal Drawing)
β	0.403	-	Chamber/Orifice diam. ratio		1.82E-01	m	
γ	1.182	-	Ratio of specific heats	Laby D(OD)	184	mm	HDGS1130-TA1-U-Final (EB Seal Drawing)
x	0.532	-	ΔP over upstream pressure		1.84E-01	m	
Y	0.355	-	Constant	Burst Disk			
ε	0.811	-	Expansion Factor	d3	4.000	in	burst disk diam - (full rupture)
		(kJ/kg)/		Nd	1.00E+00		
R	0.189	K	Gas Constant/ REFPROP	A <sub>orif</sub>	1.26E+01	in <sup>2</sup>	Orifice Area
V2	0.00362	m <sup>3</sup>	Volume of S Chamber		8.11E-03	m <sup>2</sup>	
T	700.0	C	Isothermic Temp				
	973.15	K					
P <sub>2,0</sub>	2.137	bara	Initial Pressure at burst (max burst)				
	213.7	kPa					
	2.14E+05	Pa					
Z <sub>2,0</sub>	1.000	-	Compressibility Factor/ REFPROP				
Initial Mass	0.004203	kg	Calc				



## Tool Uses:

- Transient pressure relief characteristics
- Easily and quickly work through different configurations (i.e. disk size, quantity, input pressure, laby clearance, etc.)
- Allows for more informed decision on pressure management
- Configured for versatile use cases with similar assumptions

# Conclusions

- Developed modeling tool for more in depth pressure behavior upon seal failure
  - Allowed more informed pressure mitigation strategy
- Successfully prepared test rig for hydrotesting
  - Parts quoted for machining
  - Next steps: Hydrotest the seal housing
- Progressed the project into preparation to testing phase
  - Challenges faced in motor spinning, and project management prevented testing being conducted during fellowship timeline
  - Next steps: final assembly of test rig

# Thank You

Questions